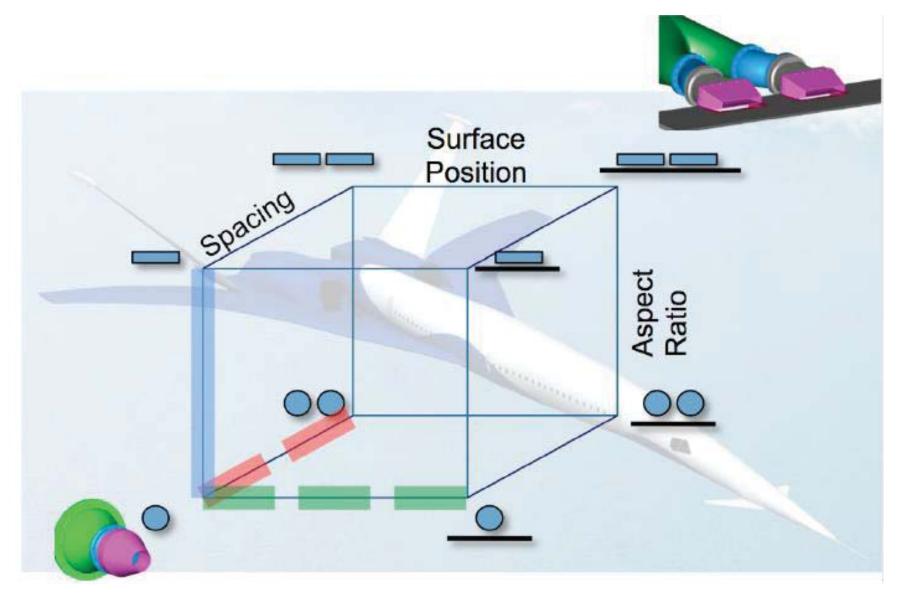


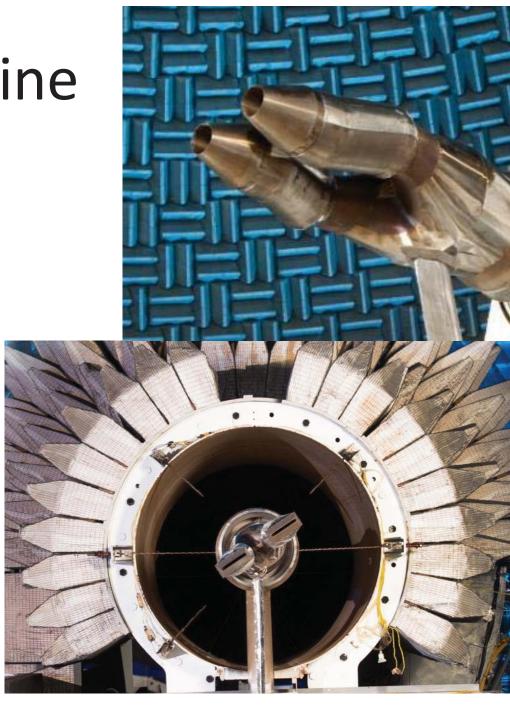
#### NASA Exhaust System Acoustic Experiments



Bridges, J., Brown, C., and Bozak, R., "Experiments on Exhaust Noise of Tightly Integrated Propulsion Systems," NASA Glenn Research Center, AIAA-2014-2904, Atlanta, GA, Jun. 2014.

#### Outline

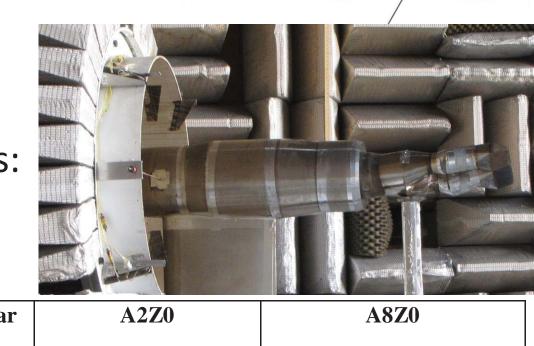
- Twin jet configurations
- Twin jet conditions
- PIV configurations
- Results
  - Single Jet Comparison
  - Twin Round Jets
  - Twin Rectangular Jets
- Summary



Twin Jet Configurations

 Jet spacing in center-to-center distance (s) over nozzle diameter (D)

- 2-inch diameter convergent round nozzles (TCON)
- 2.13-inch area equivalent diameter rectangular nozzles:
  - 2:1 aspect ratio (A2Z0)
  - 8:1 aspect ratio (A8Z0)



Y-duct

S-duct with Fairing

Twin	Round	TCON	Rectangular	A2Z0	A8Z0
<b>Spacings</b>	s/D		s/D		
<b>Z</b> 1	2.63		2.45		
Z4	3.55		3.32		
<b>Z</b> 9	5.63		5.26		

#### **Twin Jet Conditions**

#### Jet Conditions

- Acoustic Mach Number  $M_a$  from 0.7 to 1.33
- Nozzle Static Temperature Ratio  $T_{sr}$  from 0.84 to 2.27
- Forward Flight Mach Number  $M_f$  from 0.05 to 0.25

Jet Conditions		TCON				<b>A2Z0</b>	A8Z0			
$M_a$	$T_{sr}$	$M_f$	Single	<b>Z</b> 1	<b>Z</b> 4	<b>Z</b> 9	<b>Z</b> 1	Single	<b>Z</b> 4	<b>Z</b> 9
0.90	0.84	0.05		V	V	V	V		V	V
1.33	2.27	0.05		V	V	V				V
1.33	1.76	0.05		V	V	V			V	V
1.33	1.76	0.10		V						
1.33	1.76	0.20		V						
1.33	1.76	0.25	$\checkmark$	V	V	V	$\checkmark$		V	V

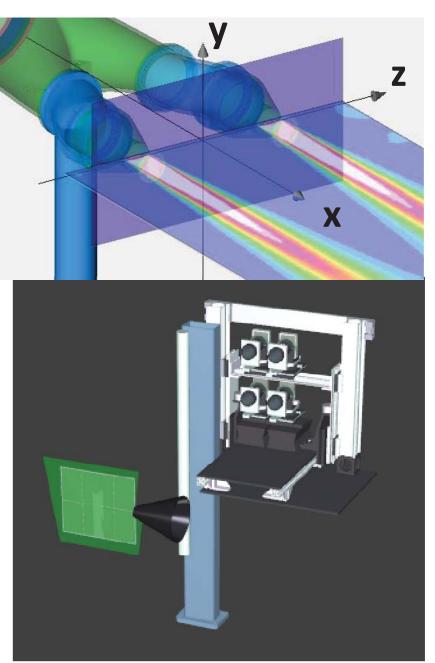
Nozzles

**Spacings** 

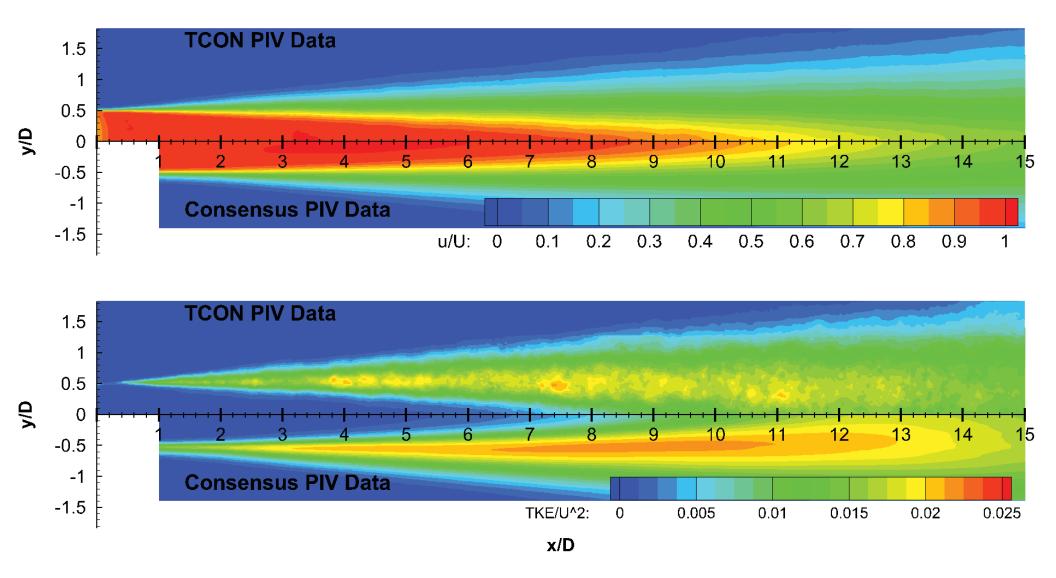
# Particle Image Velocimetry (PIV) Configurations

Coordinate system

- Streamwise (x-z plane)
  - 22" x 14" field of view
  - x/D of 0 to 30
- Cross-stream (y-z plane)
  - 10.5" x 22" field of view
  - -x/D of 1, 2, 5, 10, 15, and 20



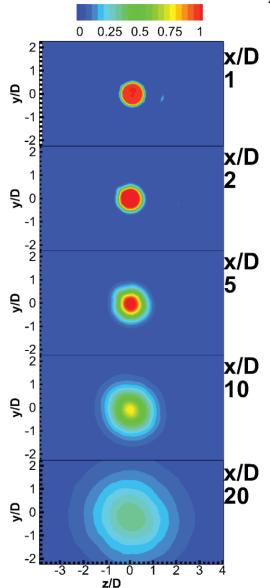
## Single Jet Comparison $M_a = 0.9$ , Unheated



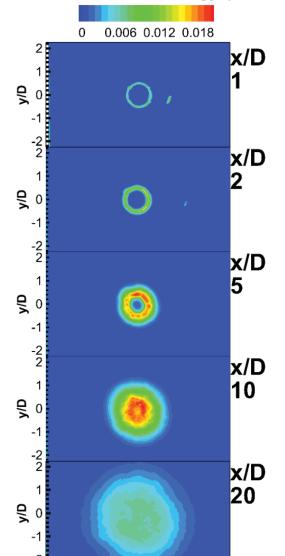
### Single Jet: Cross-stream PIV

 $M_a$  = 0.9, Unheated





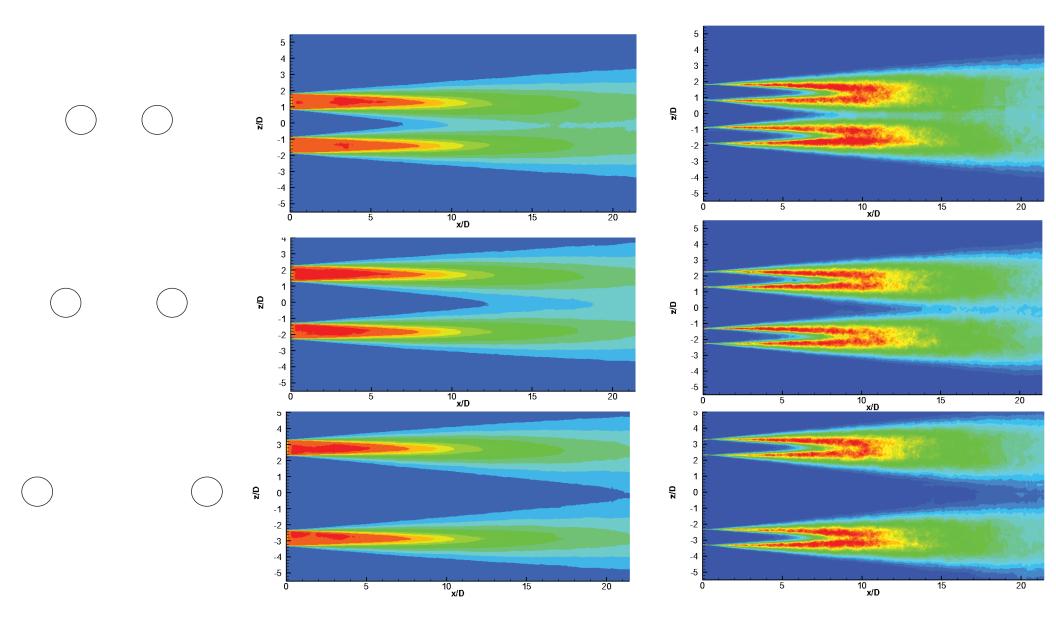




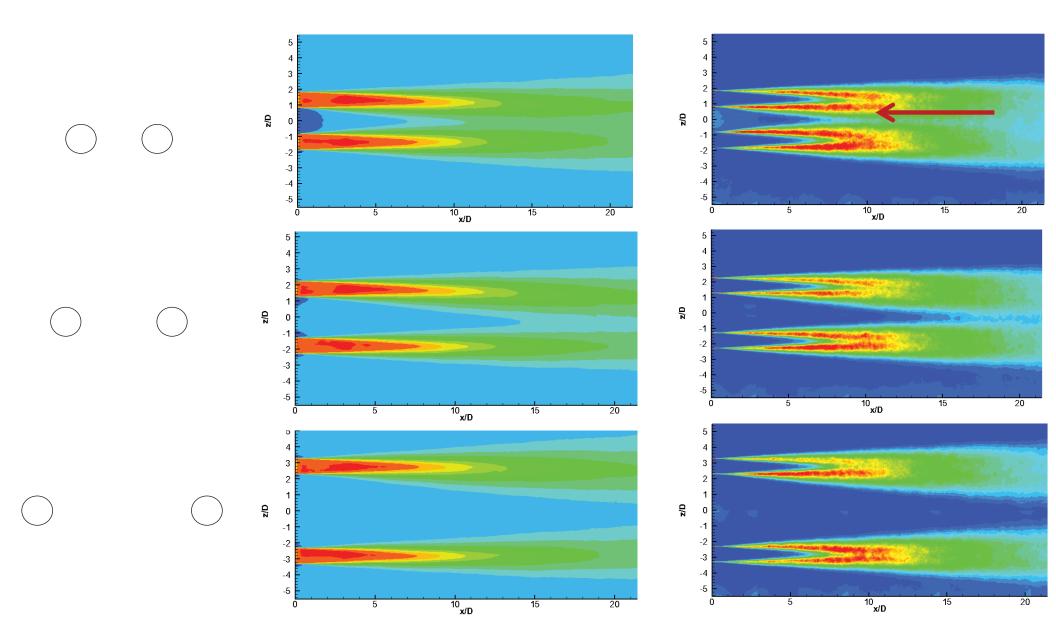
z/D



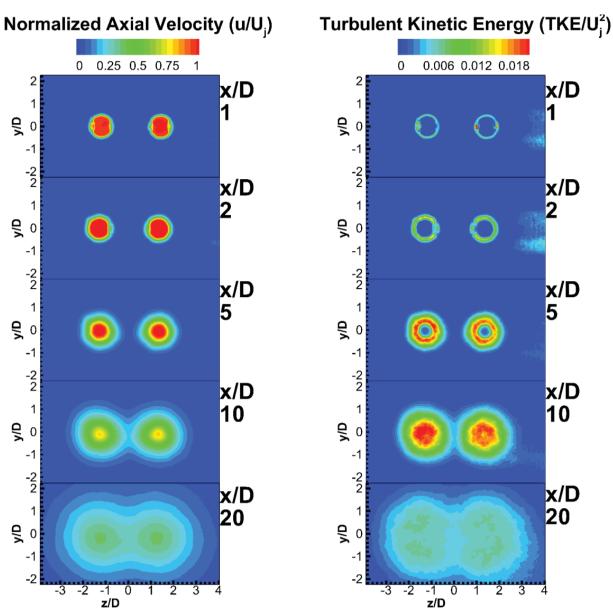
## Twin Round Jets: Streamwise PIV $M_a = 1.33$ , $T_{sr} = 1.76$ , $M_f = 0.05$



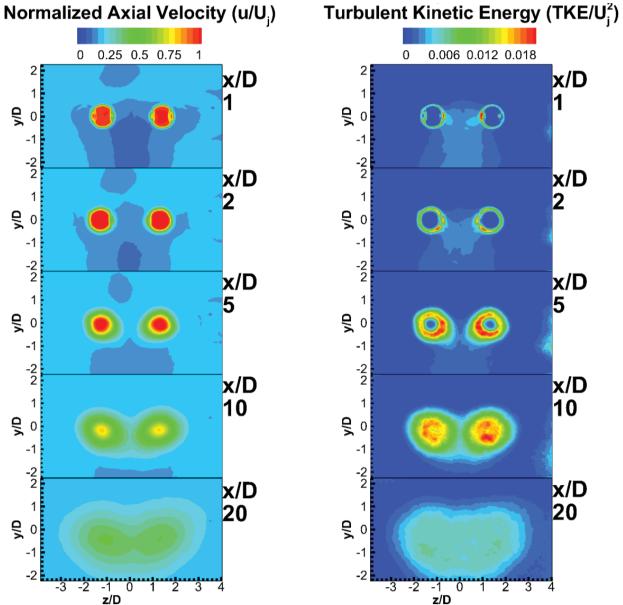
## Twin Round Jets: Streamwise PIV $M_a = 1.33$ , $T_{sr} = 1.76$ , $M_f = 0.25$



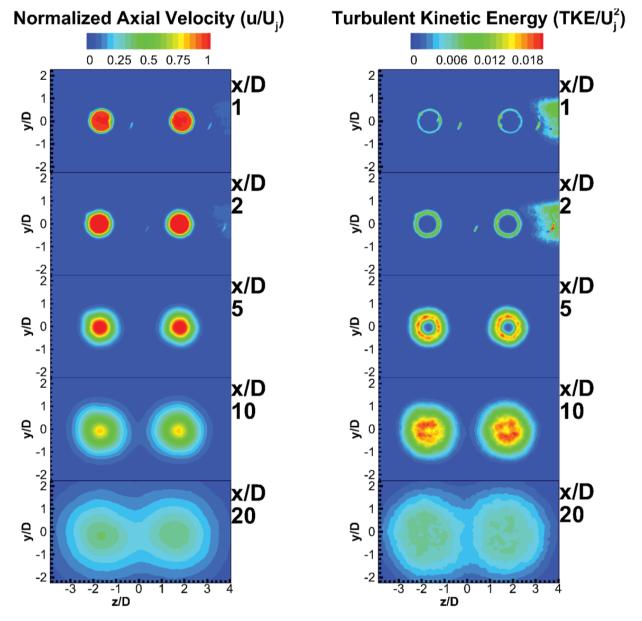
s/D = 2.63,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.05$ 



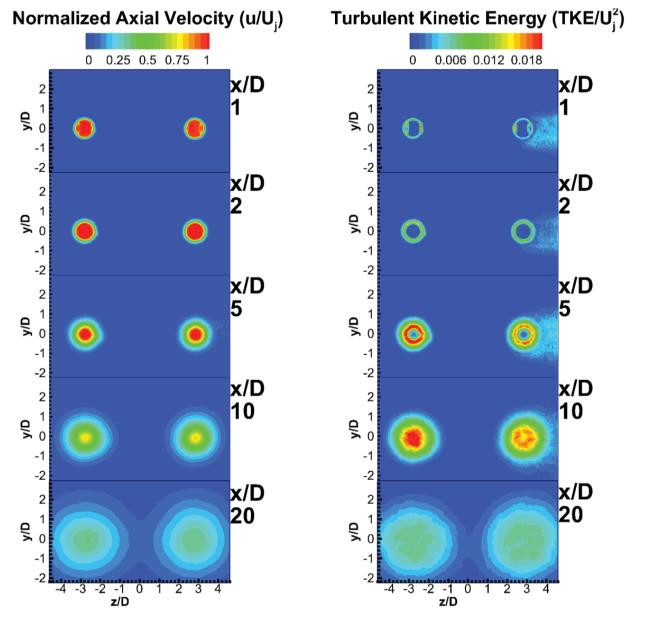
s/D = 2.63,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.25$ 



s/D = 3.55,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.05$ 

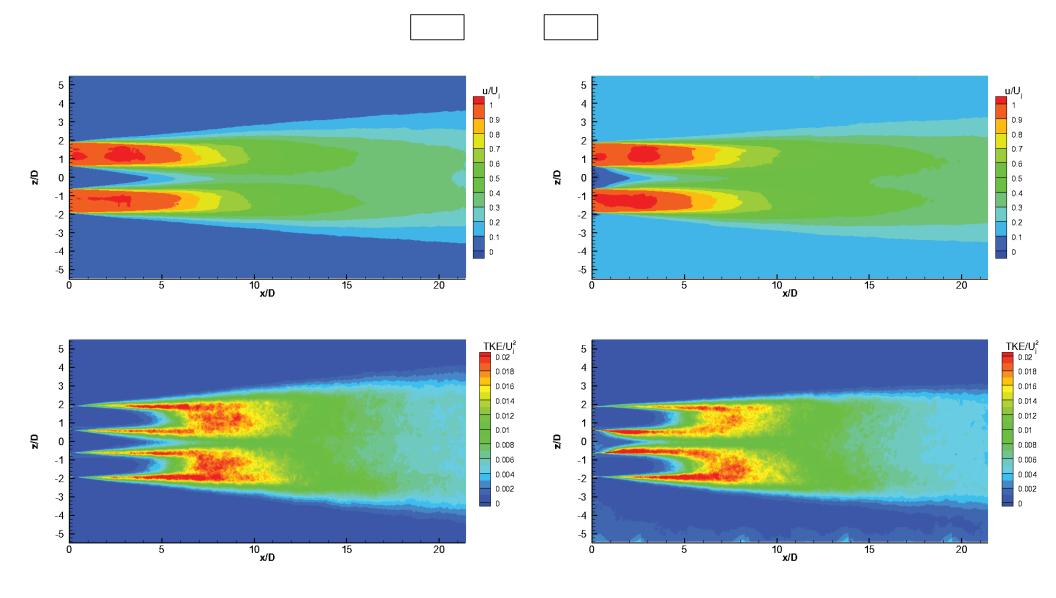


s/D = 5.63,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.05$ 



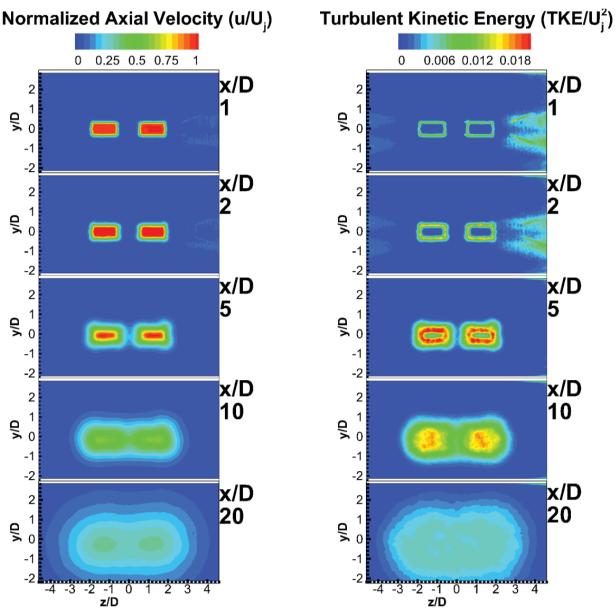
## Twin 2:1 Rectangular Jets: Streamwise PIV

$$s/D = 2.45, M_a = 1.33, T_{sr} = 1.76$$
  
 $M_f = 0.05$   $M_f = 0.25$ 



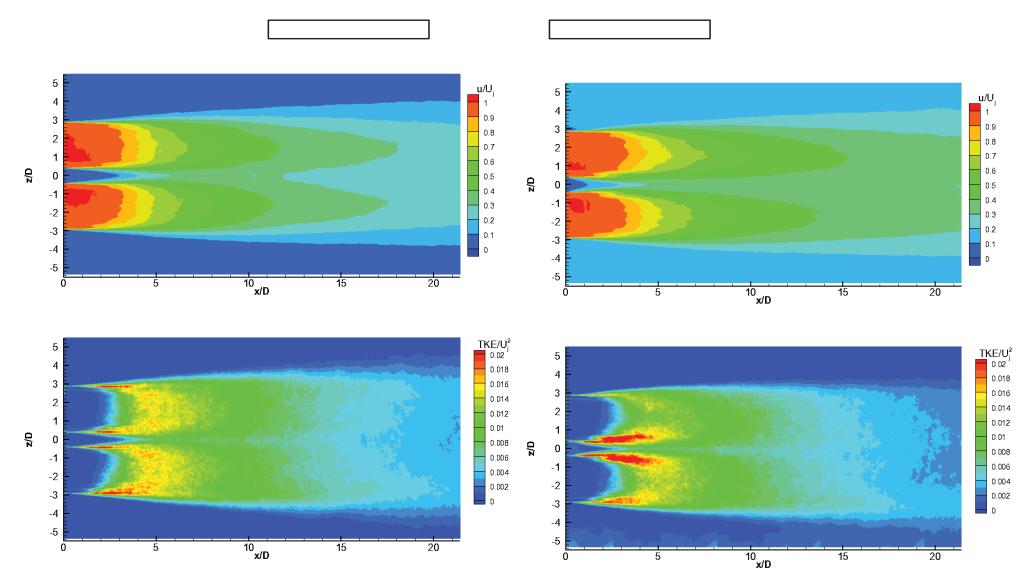
### Twin 2:1 Rectangular Jets: Cross-stream PIV

s/D = 2.45,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.05$ 



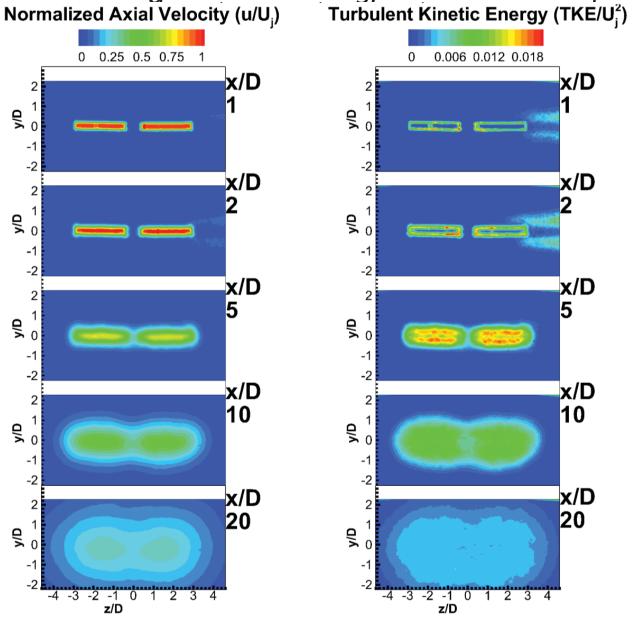
## Twin 8:1 Rectangular Jets: Streamwise PIV

$$s/D = 3.55$$
,  $M_a = 1.33$ ,  $T_{sr} = 1.76$   
 $M_f = 0.05$   $M_f = 0.25$ 



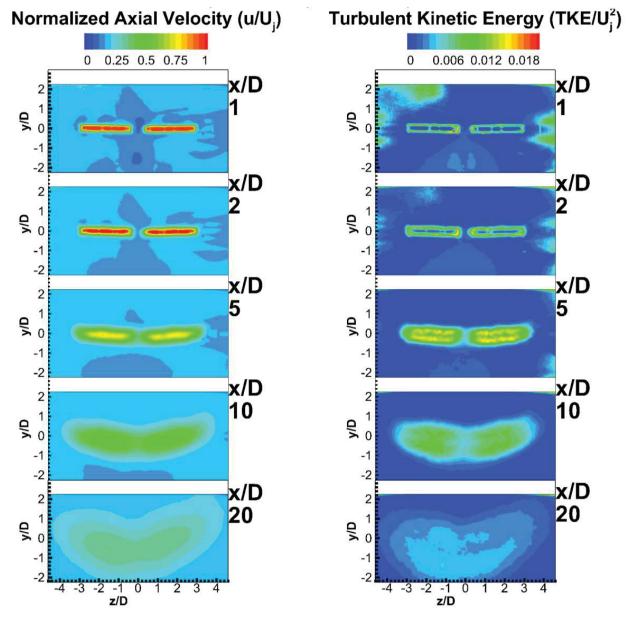
### Twin 8:1 Rectangular Jets: Cross-stream PIV

s/D = 3.55,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.05$ 

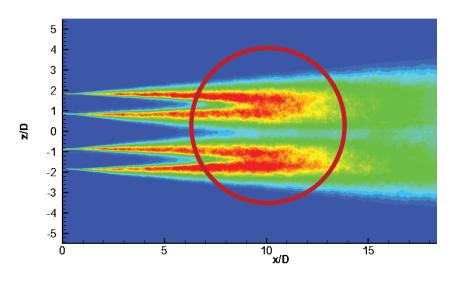


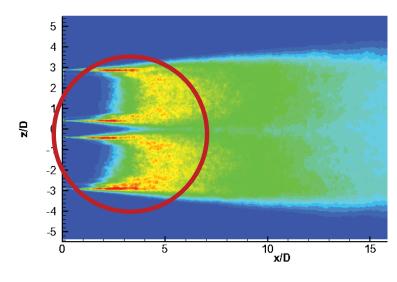
### Twin 8:1 Rectangular Jets: Cross-stream PIV

s/D = 3.55,  $M_a = 1.33$ ,  $T_{sr} = 1.76$ ,  $M_f = 0.25$ 



#### Streamwise PIV at $M_a$ = 1.33, $T_{sr}$ = 1.76, $M_f$ = 0.05 Reductions in Peak Turbulence (TKE/U<sub>i</sub><sup>2</sup>)





Twin Spacing		TCON	Rectangular (s/D)	A2Z0*	A8Z0*
<b>Z1</b>	2.63	13%	2.45	12%	X
<b>Z4</b>	3.55	0	3.32	X	10%
<b>Z9</b>	5.63	0	5.26	X	9%

<sup>\*</sup>For the rectangular nozzles, the peak turbulence does not coincide with the streamwise measurement plane.

## Summary

- Cross-stream and streamwise round and rectangular twin jet flow field PIV measurements were obtained at a few twin jet spacings and jet conditions.
- A decrease in turbulent kinetic energy levels relative to a single jet between the two jets at close spacings could be the result of enhanced mixing. This change in TKE levels is only evident in streamwise PIV measurements at the closest jet spacings obtained for each set of nozzles.
- When forward flight is increased, a velocity deficit between nozzles was measured. This appears to be due to the twin jet model blockage.
- The velocity deficit causes increased turbulence levels between the nozzles. The increased turbulence levels could be causing increased noise levels, relative to a single jet, found with far-field acoustic measurements.

## **Backup Slides**